

AD-A116 661

ARMY MATERIALS AND MECHANICS RESEARCH CENTER WATERTOWN MA F/6 11/6
DYNAMIC CHARACTERIZATION OF INTERCRITICALLY ROLLED HIGH-HARDNES--ETC(U)
JUN 82 M AZRIN, A A ANCTL, E B KULA
AMMRC-TR-82-38

UNCLASSIFIED

NL

I OF I
AD A
18661

END
DATE
FILMED
08-82
DTIC

①

AMMRC TR 82-38

AD

AD A116661

DYNAMIC CHARACTERIZATION OF INTERCRITICALLY ROLLED HIGH-HARDNESS STEEL

MORRIS AZRIN, ALBERT A. ANCTIL, and ERIC B. KULA

METALS RESEARCH DIVISION

June 1982

REC'D
JUL 8 1982
H

Approved for public release; distribution unlimited.

DTIC FILE COPY

ARMY MATERIALS AND MECHANICS RESEARCH CENTER
Watertown, Massachusetts 02172

82 07 06 256

The findings in this report are not to be construed as an official Department of the Army position, unless so designated by other authorized documents.

Mention of any trade names or manufacturers in this report shall not be construed as advertising nor as an official indorsement or approval of such products or companies by the United States Government.

DISPOSITION INSTRUCTIONS

Destroy this report when it is no longer needed.
Do not return it to the originator.

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

DTIC
ELECTE
JUL 8 1982
H

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

Block No. 20

ABSTRACT

Two compositions of intercritically rolled homogeneous steel armor were evaluated by quasi-static and dynamic tests. Cross-rolling just below the intercritical temperature resulted in improved fracture toughness at the high-hardness levels (55 and 58 HRC). Taylor cylinder ballistic impact tests indicated a dynamic stress more than double the quasi-static yield stress. The use of this test for evaluating potential high-hardness armor material is discussed.

Accession	✓
NTIS	
DTIC	
Unann	
Justif	
By	
Distrib	
Avail	
Dist	
A	



B

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

INTRODUCTION

Higher hardness in armor plate generally leads to improved ballistic performance.¹ The limiting factor at high-hardness levels is gross plate shattering, which rapidly reduces the ballistic limit. This limitation has led to the use of dual-hard armor with a high-hardness front plate and relatively ductile metallurgically roll-bonded rear plate. The high processing costs of dual-hard armor, however, have led to their replacement by specially processed rolled homogeneous steel armor.

Recently, a comprehensive study was undertaken by the U.S. Steel Corp. to develop steel compositions and processing techniques to attain high-hardness armor with adequate shatter resistance.^{2,3} For the quench-and-tempered steels it became apparent that the optimum rolling temperature was slightly below the ferrite-austenite A_3 transformation temperature. The interest in intercritical* (IC) rolling was based on earlier work on IC heat treatments that produced increased strength and fracture toughness,⁴⁻⁸ along with reduced back spalling during ballistic impact.² The resultant microstructure of reduced banding, microstructural refinement, and finely dispersed α (ferrite) and α' (martensite) regions is desirable in terms of ballistic performance.

A natural extension of IC heat treatments is the use of IC rolling to further refine the microstructure. This is possible since the low temperatures used essentially prevent recrystallization. The two hypoeutectoid steels (Table 1) that are the subject of this study were obtained from the U.S. Steel Corp. The transformation and processing temperatures are shown in Table 2. A number of microstructural features are a direct result of IC rolling. Intercritical holding time is also a factor since a high-carbon austenite results from the α -phase rejection of carbon. The finely dispersed α -regions are refined during IC rolling, producing a layered microstructure of ferrite and austenite. On quenching, after IC rolling, α' forms in the austenite and the resulting retained austenite and banded $\alpha+\alpha'$ would be expected to provide improved longitudinal fracture toughness. This microstructure can be undesirable, however, in armor applications where failure by delamination is strongly influenced by microstructural layering. On the other hand, refining the microstructure by thermomechanical treatments should result in spalling resistance at least equivalent to that of conventionally processed rolled homogeneous armor. Crystallographic preferred orientations produced in the austenite at high temperatures are not destroyed on quenching. The high carbon γ (austenite) transforms to high carbon α' .

*The intercritical region is the two-phase α - γ region bounded by A_1 and A_3 .

1. MANGANELLO, S. J., and ABBOTT, K. H. *Metallurgical Factors Affecting the Ballistic Behavior of Steel Targets*. J. of Materials, v. 7, no. 2, June 1972, p. 231-239.
2. CARSON, C. G., DABKOWSKI, D. S., SPAEDER, G. J., and PORTER, L. F. *A Research Study on the Relative Merits of Homogeneous and Dual-Hardness Armor Produced by Special Processes (U)*. U.S. Steel Corp., Contract DAAG46-71-C-0136, Final Report, AMMRC CTR 72-14, September 1972, AD 522367 (Confidential Report).
3. SPEICH, G. R., HU, H., and MILLER, R. L. *Effect of Preferred Orientation and Related Metallurgical Parameters on Mechanical Properties and Ballistic Performance of High-Hardness Steel Armor*. U.S. Steel Corp., Contract DAAG46-73-C-0244, Final Report, AMMRC CTR 74-39, June 1974.
4. DULIEU, D., LATHAM, D. L., BANNISTER, J. W., and GIBSON, S. *Controlled Rolling of Carbon and Low Alloy Steel*. BISRA Open Report, MG/INT/73/70, British Steel Corp., 1970.
5. GRANGE, R. A. *Fibrous Microstructures Developed in Steel by Thermomechanical Processing*. Proc. Second International Conference on the Strength of Metals and Alloys, ASM, Menlo Park, Ohio, 1970, p. 861-876.
6. BERNSTEIN, M. L., ODESSKII, P. D., and KORNEEVA, G. B. *Thermomechanical Treatment of Low-Alloy Steels by Deformation in the Intercritical Range*. Steel in the USSR, v. 2, no. 11, 1972, p. 914-916.
7. MARCHENKO, V. G. *Quenching of Steels from Intercritical Temperatures*. Metal Science and Heat Treatment, v. 17, March-April 1975, p. 245-246.
8. WADA, T., and DOANE, D. V. *The Effect of an Intercritical Heat Treatment on Temper Embrittlement of a Ni-Cr-Mo-V Rotor Steel*. Met. Trans., v. 5, 1974, p. 231-239.

Quenching from below A_3 reduces the amount of retained γ . These factors are known to influence strength, toughness, and the resultant penetration resistance.¹⁻³ Carson et al.² found that quench-and-tempered IC rolled homogeneous armor plate had improved ballistic performance without plate shattering.

Table 1. CHEMICAL COMPOSITION, WEIGHT PERCENT*

Material	Mn	P	S	Si	Ni	Cr	Mo	Al	N	V	Cu
0.39C	0.60	0.005	0.007	1.29	5.46	--	0.49	0.031	0.009	0.10	0.94
0.47C	0.66	0.006	0.007	0.32	1.10	0.76	0.51	0.024	0.008	0.26	--

*Material obtained from U.S. Steel Corp. in the processed condition.

Table 2. TRANSFORMATION AND PROCESSING TEMPERATURES*

Material	Austenitizing Temp. (°F)	Transformation Temp.* (°F)		Rolling Temp. (°F)	HRC
		A_1	A_3		
0.39C	2200	1225	1440	1365	56
0.47C	1800	1325	1495	1420	58

*Calculated values (Ref. 2).

The two compositions in Table 1 were studied to determine the dynamic characteristics important to armor applications. In addition to tensile and fracture toughness testing, Taylor projectile impact tests were also conducted.^{9,10} Material characterization requires that test conditions be relevant to those encountered in service. This is particularly true for armor applications where strain rates above 10^4 sec^{-1} are encountered. The Taylor impact test relates the length change of an impacted flat-end projectile to a dynamic flow stress. The test procedure is relatively simple. The projectile is fired at a right angle to a rigid, thick target. Low impact velocities are used to prevent fracture. Contrary to expectations, the measured Taylor stress of moderate strength materials is nearly independent of projectile impact velocity.¹⁰⁻¹³ These expectations are based on tension and compression results in the 10^{-4} to 10^3 sec^{-1} strain rate range. However, a collection of tension and compression results by Soohoo et al.¹⁴ show the strain rate dependence of yield strength, though significant at low- and intermediate-strength levels, becomes negligible at the high strength levels.

Measurement of the impact velocity and deformed projectile length permit calculation of the Taylor dynamic flow stress Y^0 from the equation:

9. TAYLOR, G. I. *The Use of Flat-Ended Projectiles for Determining Dynamic Yield Stress. I: Theoretical Considerations.* Proc. Royal Soc., v. A 194, 1948, p. 289-299.
10. WILKINS, M. L., and GUINAN, M. W. *Impact of Cylinders on a Rigid Boundary.* J. Appl. Phys., v. 44, no. 3, 1973, p. 1200-1206.
11. WHIFFIN, A. C. *The Use of Flat-Ended Projectiles for Determining Dynamic Yield Stress. II: Tests on Various Metallic Materials.* Proc. Royal Soc., v. A 194, 1948, p. 300-322.
12. LEE, E. H., and TUPPER, S. J. *Analysis of Plastic Deformation in a Steel Cylinder Striking a Rigid Target.* J. Appl. Mech., v. 21, 1954, p. 63-70.
13. KARNES, C. W., and BERTHOLF, L. D. *Inelastic Behavior of Solids.* McGraw-Hill, New York, 1970.
14. SOOHOO, P., JIANG, C. W., and CHEN, M. M. *Dynamic Properties of Materials, Part III - Steels.* Boston University, Contract DAAG46-73-C-0181, Final Report, AMMRC CTR 74-24, April 1974.

$$\frac{L_f}{L_o} = \exp \left[- \frac{\rho V^2}{2Y^0} \right], \quad (1)$$

L_f = final projectile length,
 L_o = original projectile length,
 ρ = density, and
 V = impact velocity.

In Equation 1 Y^0 represents an "average" flow stress that a cylindrical rod experiences as it decelerates and deforms into a mushroom-shaped rod. The length measurement is made after the impacted end of the projectile undergoes gross deformation with negligible, if any, deformation at the opposite end of the projectile. Therefore, Y^0 represents neither a yield stress nor ultimate stress (i.e., stress at maximum load).

Tests by Taylor⁹ and others,^{10-13,15} were performed using low strength materials. In all cases, Y^0 was higher than the quasistatic value. More recently, Papirno et al.¹⁶ applied the Taylor test to high-strength 4340 steel and found that the dynamic stress was almost double the static compression yield stress.

MATERIALS

The rolled homogeneous armor steels tested (Table 1) were received in the IC rolled-and-tempered condition. The higher carbon alloy is a standard armor steel (0.50C-1.1Ni-0.75Cr-0.50Mo) modified with 0.2 percent vanadium. Although the alloys were intercritically cross-rolled with a final rolling ratio of one-to-one, there was still microstructural evidence of a "rolling direction." All mechanical test specimens were oriented with reference to the apparent longitudinal direction of the 5/8-inch-thick plates.

Figure 1 shows the heavily banded structure for both alloys. Areas (dark) of dense carbide precipitation are also evident. At higher magnification, the 0.47C alloy has a finer lath martensite packet size. Rolling conditions and quench rates produced no noticeable difference in microstructure at the surface and midsection planes (Figure 2). The 0.47C alloy has a finer grain structure with carbides strung out along prior austenite grain boundaries. The bands appear as patches when viewed normal to the plate (Figure 2).

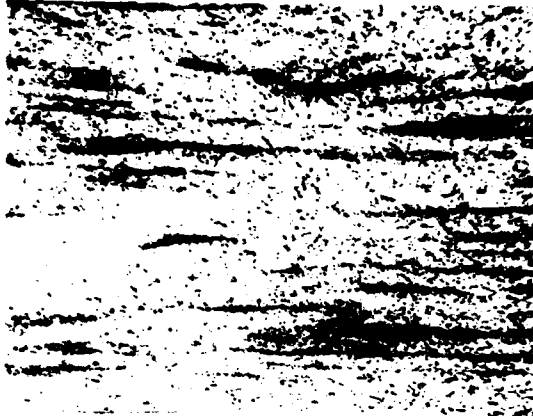
EXPERIMENTAL PROCEDURES

Selected room temperature tests were performed to complement those reported by the U.S. Steel Corp. Research Laboratory.^{2,3} Static tension tests were performed parallel to the plate rolling direction. Computerized instrumented Charpy testing equipment was used to obtain the dynamic fracture toughness data. Charpy impact and dynamic fracture toughness tests were performed in the LT and TL directions.

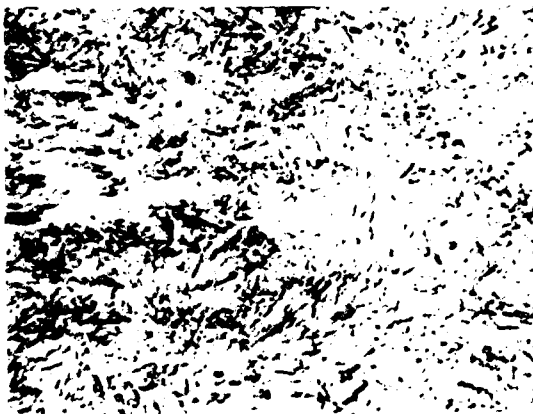
15. POLOSATKIN, G. D., KUDRYAVTSEVA, L. A. and GLAZKOV, V. M. *Russian Metallurgy*, no. 5, 1966, p. 62-64.

16. PAPIRNO, R. P., MESCALL, J. F., and HANSEN, A. M. *Proceedings of the Army Symposium on Solid Mechanics, 1980 - Designing for Extremes: Environment, Loading, and Structural Behavior*. Army Materials and Mechanics Research Center, AMMRC MS 80-4, September 1980, p. 367-385.

Taylor impact projectiles, 0.218-inch diameter and 0.436-inch long, were machined from a 5/8-inch plate with the specimen axis in the short transverse and longitudinal directions. The projectiles were fired from a 0.218-inch-diameter smooth bore light gas gun at a thick hardened-steel plate. Precautions were taken to ensure normal impact and accurate final length (L_f) measurements.* Velocities were measured with a pair of silver-coated paper screens located close to the target.



Magnification 100X



(a)

Magnification 1000X

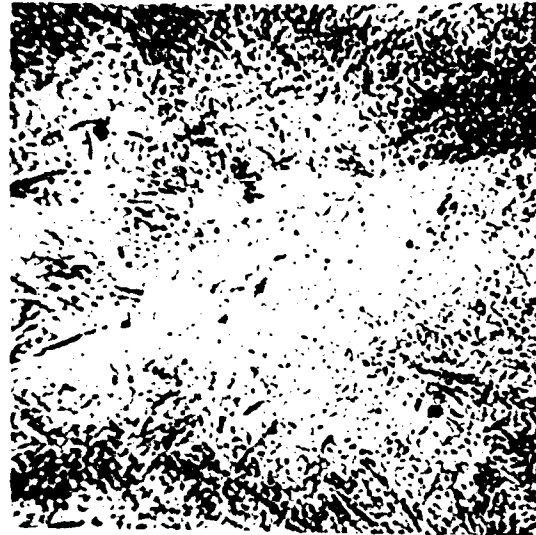
Figure 1. Photomicrographs of the plane normal to the rolling direction (a) for the 0.47C and (b) for the 0.39C homogeneous intercritically rolled steel armor. Picral etch.

*PAPIRNO, et al.¹⁶ clearly demonstrated that for high-strength materials, where length shortening is small, the uncertainty in L_f can result in a significant error in the calculated Taylor stress.

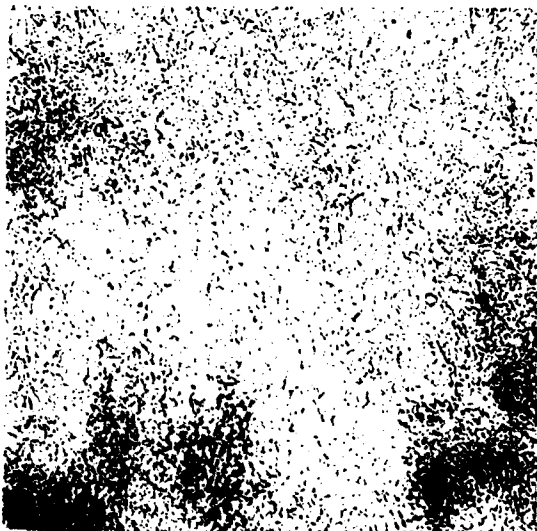
0.47C



0.39C



Surface



(a)



(b)

Midthickness

Figure 2. Photomicrographs of the plane parallel to the rolling direction (a) outer surface plane (b) plane at midthickness of homogeneous intercritically rolled homogeneous armor. Picral etch. Mag. 100X.

RESULTS AND DISCUSSION

The 0.47C steel, as expected, had higher hardness and static strength levels (Tables 2 and 3), lower ductility (Table 3), and lower Charpy impact energy (Table 4). The lack of orientation effects for both steels are a result of cross-rolling. The Charpy energy levels are slightly higher than other high strength steels.¹⁷ Apparently, at these moderate strain rates, IC rolling and the structural refinement are beneficial. Similar results are found for the dynamic fracture toughness specimens (Table 5). Again, fracture toughness is lower at the higher strength level, orientation plays no significant role, and values reported are higher than for comparable high strength steels.¹⁸

Table 3. LONGITUDINAL TENSILE PROPERTIES

Material	0.2% YS (ksi)	UTS (ksi)	TFS (ksi)	n	Elong. (%)	RA (%)
0.39C	226	320	409	-	11	29
	226	317	392	0.09	9	25
	240	317	401	0.09	11	29
0.47C	266	351	411	-	7	20
	275	353	415	0.08	9	20
	-	355	420	0.07	7	21

TFS - True Fracture Stress

n - Strain Hardening Exponent

Table 4. CHARPY IMPACT ENERGY

Material	Orientation	Energy ft-lb
0.39C	LT	19.4
	LT	15.2
	TL	15.5
	TL	17.8
0.47C	LT	10.3
	LT	12.1
	TL	11.2
	TL	11.2

Table 5. DYNAMIC FRACTURE TOUGHNESS

Material	Orientation	K_{I0} (ksi $\sqrt{in.}$)
0.39C	LT	62.3
	TL	73.2
0.47C	LT	53.1
	TL	52.6

17. TUFFNELL, G. W., and CAIRNS, R. L. 18% Nickel 350-Mnaging Steel. Trans. ASM, v. 61, 1968, p. 798-806.

18. WOOD, W. E., PARKER, E. R., and ZACKAY, V. F. An Investigation of Metallurgical Factors Which Affect Fracture Toughness of Ultra-High Strength Steels. University of California, Contract DAAG46-72-C-0220, Final Report, AMMRC CTR 73-24, May 1973.

AD
Army Materials and Mechanics Research Center,
Watertown, Massachusetts 02172
DYNAMIC CHARACTERIZATION OF INTERCRITICALLY
ROLLED HIGH-HARDNESS STEEL - Morris Azrin,
Albert A. Anctil, and Eric B. Kula

Technical Report AMRC TR 82-38, June 1982, 11 pp -
illus-tables, D/A Project IL161102AH42,
AMCMS Code 611102.H420011

Two compositions of intercritically rolled homogeneous steel armor were evaluated by quasi-static and dynamic tests. Cross-rolling just below the intercritical temperature resulted in improved fracture toughness at the high-hardness levels (55 and 58 HRC). Taylor cylinder ballistic impact tests indicated a dynamic stress more than double the quasi-static yield stress. The use of this test for evaluating potential high-hardness armor material is discussed.

AD
UNCLASSIFIED
UNLIMITED DISTRIBUTION
Key Words

Dynamic properties
High strength steels
Armor

AD
Army Materials and Mechanics Research Center,
Watertown, Massachusetts 02172
DYNAMIC CHARACTERIZATION OF INTERCRITICALLY
ROLLED HIGH-HARDNESS STEEL - Morris Azrin,
Albert A. Anctil, and Eric B. Kula

Technical Report AMRC TR 82-38, June 1982, 11 pp -
illus-tables, D/A Project IL161102AH42,
AMCMS Code 611102.H420011

Two compositions of intercritically rolled homogeneous steel armor were evaluated by quasi-static and dynamic tests. Cross-rolling just below the intercritical temperature resulted in improved fracture toughness at the high-hardness levels (55 and 58 HRC). Taylor cylinder ballistic impact tests indicated a dynamic stress more than double the quasi-static yield stress. The use of this test for evaluating potential high-hardness armor material is discussed.

AD
UNCLASSIFIED
UNLIMITED DISTRIBUTION
Key Words

Dynamic properties
High strength steels
Armor

AD
Army Materials and Mechanics Research Center,
Watertown, Massachusetts 02172
DYNAMIC CHARACTERIZATION OF INTERCRITICALLY
ROLLED HIGH-HARDNESS STEEL - Morris Azrin,
Albert A. Anctil, and Eric B. Kula

Technical Report AMRC TR 82-38, June 1982, 11 pp -
illus-tables, D/A Project IL161102AH42,
AMCMS Code 611102.H420011

Two compositions of intercritically rolled homogeneous steel armor were evaluated by quasi-static and dynamic tests. Cross-rolling just below the intercritical temperature resulted in improved fracture toughness at the high-hardness levels (55 and 58 HRC). Taylor cylinder ballistic impact tests indicated a dynamic stress more than double the quasi-static yield stress. The use of this test for evaluating potential high-hardness armor material is discussed.

AD
UNCLASSIFIED
UNLIMITED DISTRIBUTION
Key Words

Dynamic properties
High strength steels
Armor

AD
Army Materials and Mechanics Research Center,
Watertown, Massachusetts 02172
DYNAMIC CHARACTERIZATION OF INTERCRITICALLY
ROLLED HIGH-HARDNESS STEEL - Morris Azrin,
Albert A. Anctil, and Eric B. Kula

Technical Report AMRC TR 82-38, June 1982, 11 pp -
illus-tables, D/A Project IL161102AH42,
AMCMS Code 611102.H420011

Two compositions of intercritically rolled homogeneous steel armor were evaluated by quasi-static and dynamic tests. Cross-rolling just below the intercritical temperature resulted in improved fracture toughness at the high-hardness levels (55 and 58 HRC). Taylor cylinder ballistic impact tests indicated a dynamic stress more than double the quasi-static yield stress. The use of this test for evaluating potential high-hardness armor material is discussed.

AD
UNCLASSIFIED
UNLIMITED DISTRIBUTION
Key Words

Dynamic properties
High strength steels
Armor

Table 6 lists the cylinder impact test results along with specimen orientations. There is a lower and upper velocity limit that can be used for these tests. The higher velocities produced excessive deformation resulting in specimen fracture along a 45° shear plane (Figure 3). This observation is consistent with the reported cone-shaped fracture observed in cylinder impact tests of high strength 4340 steel.¹⁶ At lower velocities (not shown in Table 6) length shortening was insignificant. The impact data suitable for Taylor model calculations are shown with an asterisk. The harder, higher carbon alloy has the higher dynamic Taylor stress Y^0 while both steels show no strong orientation effect. The Y^0 is more than double the static 0.2 percent yield stress.

Table 6. CYLINDRICAL IMPACT DATA

Material	Orientation	Velocity (ft/sec)	Original Length L_0 (in.)	Final Length L_f (in.)	L_f/L_0	Y^0 (ksi)	Impact Observation
0.39C	Short Transverse	640	0.440	0.424	0.963	578	Shear Cracking - No separation
		697*	0.441	0.422	0.956		Deformation - No cracking
		706	0.441	-	-		Shear Fracture
		730	0.439	-	-		Shear Fracture
		735	0.441	0.418	0.947		Shear Cracking - No separation
		776	0.442	-	-		Shear Fracture
		820	0.440	-	-		Shear Fracture
	Longitudinal	630	0.440	-	-	578	Shear Fracture
		635	0.440	0.425	0.966		Shear Cracking - No separation
		653	0.441	-	-		Shear Fracture
		654	0.441	0.423	0.959		Shear Cracking - No separation
		657	0.441	-	-		Shear Fracture
		658	0.442	-	-		Shear Fracture
		658*	0.442	0.425	0.962		Deformation - No cracking
		672	0.441	-	-		Shear Fracture
0.47C	Short Transverse	637	0.441	0.427	0.968	724	Shear Cracking - No separation
		639	0.440	-	-		Shear Fracture
		643*	0.441	0.428	0.969		Deformation - No cracking
		648	0.439	-	-		Shear Fracture
		648	0.441	-	-		Shear Fracture
		649	0.440	-	-		Shear Fracture
		651	0.442	-	-		Shear Fracture
		668*	0.440	0.426	0.968		Deformation - No cracking
	Longitudinal	607*	0.440	0.427	0.969	643	Deformation - No cracking
		607	0.441	0.429	0.973		Deformation - Crack initiation
		609	0.441	-	-		Shear Fracture
		611	0.441	0.430	0.974		Deformation - Crack initiation
		616	0.439	-	-		Shear Fracture
		645*	0.441	0.428	0.970		Deformation - No cracking
		650	0.438	0.424	0.968		Shear Cracking - No separation
		650	0.440	-	-		Shear Fracture

*Data used in Taylor Calculation

The quasi-static yield, Charpy impact, and Taylor impact results are summarized in Figure 4 and compared with published data for quenched-and-tempered AISI 4340 steel.^{16,19-25} The Y^0 band¹⁶ represents two L/D ratios, the L/D = 2 ratio generally having higher values than L/D = 4. The most significant observation is the high value of Y^0 for the IC rolled material. At the high-hardness level Y^0 for the 4340 steel is double the quasi-static tensile yield strength. This ratio is approximately 2-1/2 for the IC rolled material. The reason for the superior performance of the intercritically rolled material is not known.

The experimental simplicity and ease of calculations make the Taylor test desirable for evaluating potential armor materials. A high dynamic Taylor stress (actually an approximate average flow stress) would give some guidance in the search for increasing hardness and impact resistance. Unfortunately, no studies have been reported relating armor performance and the calculated Y^0 . In addition, there is only a small body of literature comparing quasi-static and Taylor impact tests for low-strength materials. Only recently have results been reported for high-strength alloys.¹⁶ Based on experimental and theoretical analysis, Papirno et al.¹⁶ concluded that Equation 1 is nonconservative when applied to high strength steels. Also included in that paper is a discussion of the experimental sophistication necessary for the Taylor test.

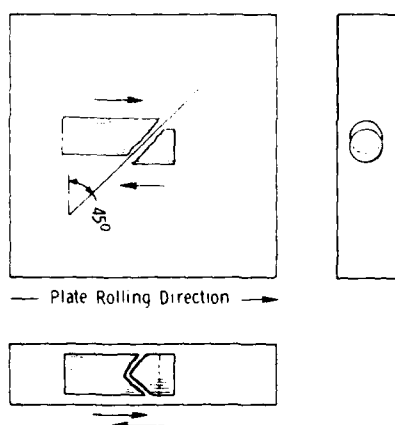


Figure 3. Cylindrical fracture plane.

19. KLINGER, L. J., BARNETT, W. J., FROHMBERG, R. P., and TROIANO, A. R. *The Embrittlement of Alloy Steel at High Strength Levels*. Trans. ASM, v. 46, 1954, p. 1557-1598.
20. Allegheny Ludlum Steel Corp., Extrusion Laboratory. *Mechanical Test Results on Various Extruded Materials*, January 1956.
21. LARSON, F. R., and NUNES, J. *Relationship Between Energy, Fibrosity, and Temperature in Charpy Impact Tests on AISI 4340 Steel*. Army Materials and Mechanics Research Center, WAL-TR-834.2/3, December 1961, AD 269779.
22. FITZGIBBON, D. P. *Semiannual Report on Pressure Vessel Design Criteria*. Space Technology Laboratories, Inc., Air Force Ballistic Missile Division, TR-59-0000-00714, June 1959, AD 607630.
23. FREEDMAN, A. H. *An Accelerated Stress Corrosion Test for High-Strength Ferrous Alloys*. J. of Materials, v. 5, no. 2, 1970, p. 431-466.
24. *Properties and Selection: Iron and Steel*. ASM Metals Handbook, 9th Edition, v. 1, 1978, p. 426.
25. KULA, E. B., and ANCTIL, A. A. *Tempered Martensite Embrittlement and Fracture Toughness in SAE 4340 Steel*. J. of Materials, v. 4, no. 4, 1969, p. 817-841.

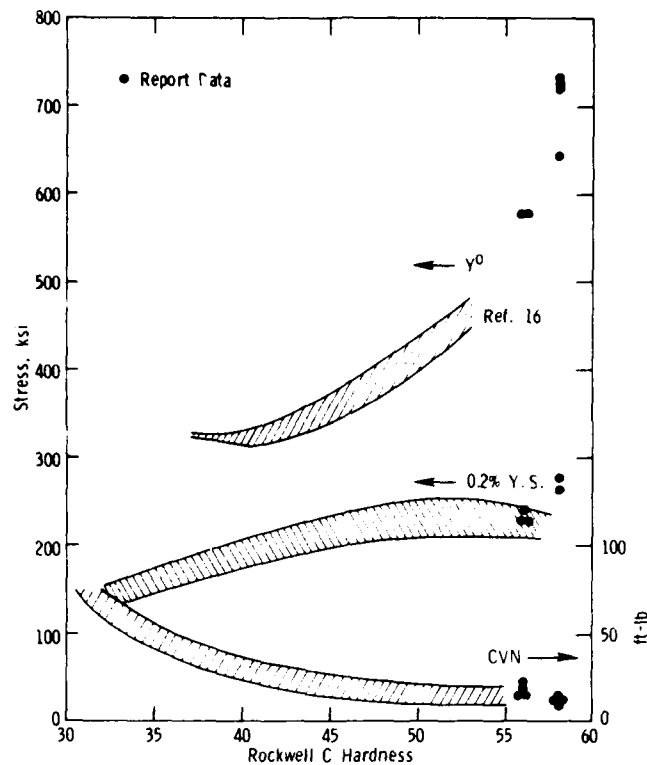


Figure 4. Taylor stress, tensile yield stress, and Charpy impact energy of quenched-and-tempered AISI 4340 steel (shaded areas, Ref. 16, 19-25) and intercritically rolled steel.

CONCLUSIONS

1. Intercritically rolled homogeneous armor steel has higher strength and toughness than armor processed by a conventional quench-and-temper treatment.
2. The dynamic Taylor flow stress Y_0 , obtained from the cylinder impact test, is more than double the quasi-static tensile yield strength of the two steels tested.
3. The relatively simple Taylor impact test is a potential method of evaluating candidate armor materials.

ACKNOWLEDGMENTS

The authors thank Dr. Gregory B. Olson and John Mescall for helpful discussions during the preparation of this report.

DISTRIBUTION LIST

No. of
Copies

To

- 1 Office of the Under Secretary of Defense for Research and Engineering,
The Pentagon, Washington, DC 20301
- 12 Commander, Defense Technical Information Center, Cameron Station,
Building 5, 5010 Duke Street, Alexandria, VA 22314

Metals and Ceramics Information Center, Battelle Columbus Laboratories,
505 King Avenue, Columbus, OH 43201
- 1 ATTN: J. H. Brown, Jr.

Deputy Chief of Staff, Research, Development, and Acquisition,
Headquarters, Department of the Army, Washington, DC 20310
- 1 ATTN: DAMA-ARZ

Commander, Army Research Office, P.O. Box 12211, Research
Triangle Park, NC 27709
- 1 ATTN: Information Processing Office

Commander, U.S. Army Materiel Development and Readiness Command,
5001 Eisenhower Avenue, Alexandria, VA 22333
- 1 ATTN: DRCLDC

Commander, U.S. Army Materiel Systems Analysis Activity,
Aberdeen Proving Ground, MD 21005
- 1 ATTN: DRXSY-MP, Director

Commander, U.S. Army Missile Command, Redstone Arsenal, AL 35809
- 1 ATTN: Technical Library
- 1 DRSMI-CS, R. B. Clem

Commander, U.S. Army Armament Research and Development Command, Dover,
NJ 07801
- 2 ATTN: Technical Library
- 1 DRDAR-SCM, J. D. Corrie
- 1 Dr. J. Waldman

Commander, U.S. Army Tank-Automotive Command, Warren, MI 48090
- 1 ATTN: DRSTA-RKA
- 2 DRSTA-UL, Technical Library
- 1 DRSTA-RCK, Dr. J. Chevalier

Commander, U.S. Army Foreign Science and Technology Center,
220 7th Street, N.E., Charlottesville, VA 22901
- 1 ATTN: Military Tech, Mr. Marley

Director, Eustis Directorate, U.S. Army Air Mobility Research and
Development Laboratory, Fort Eustis, VA 23604
- 1 ATTN: DAVDL-E-MOS
- 1 DAVDL-EU-TAP

No. of
Copies

To

U.S. Army Aviation Training Library, Fort Rucker, AL 36360
1 ATTN: Building 5906--5907

Commander, U.S. Army Aviation Research and Development Command,
4300 Goodfellow Boulevard, St. Louis, MO 63120
1 ATTN: DRDAV-EGX
1 DRDAV-EX, Mr. R. Lewis
1 DRDAV-EQ, Mr. Crawford
1 DRCPM-AAH-TM, Mr. R. Hubbard
1 DRDAV-DS, Mr. W. McClane

Naval Research Laboratory, Washington, DC 20375
1 ATTN: Dr J. M. Krafft - Code 5830
1 Code 2627

Chief of Naval Research, Arlington, VA 22217
1 ATTN: Code 471

Director, Structural Mechanics Research, Office of Naval Research,
800 North Quincy Street, Arlington, VA 22203
1 ATTN: Dr. N. Perrone

Commander, U.S. Air Force Wright Aeronautical Laboratories,
Wright-Patterson Air Force Base, OH 45433
2 ATTN: AFWAL/MLSE, E. Morrissey
1 AFWAL/MLC
1 AFWAL/MLLP, D. M. Forney, Jr.
1 AFWAL/MLBC, Mr. Stanley Schulman
1 AFWAL/MLXE, A. Olevitch

National Aeronautics and Space Administration, Washington, DC 20546
1 ATTN: Mr. B. G. Achhammer
1 Mr. G. C. Deutsch - Code RW

National Aeronautics and Space Administration, Marshall Space Flight
Center, Huntsville, AL 35812
1 ATTN: R. J. Schwinghammer, EH01, Dir, M&P Lab
1 Mr. W. A. Wilson, EH41, Bldg. 4612

Chief of Naval Operations, Washington, DC 20350
1 ATTN: OP-987, Director

Aeronautical Systems Division (AFSC), Wright-Patterson Air Force Base,
OH 45433
1 ATTN: ASD/ENFEF, D. C. Wight
1 ASD/ENFTV, D. J. Wallick
1 ASD/XRHD, G. B. Bennett

Air Force Armament Laboratory, Eglin Air Force Base, FL 32542
1 ATTN: AFATL/DLYA, V. D. Thornton

No of
Copies

To

Air Force Flight Dynamics Laboratory, Wright-Patterson Air Force Base, OH 45433

- 1 ATTN: AFFDL/FES, G. W. Ducker
- 1 AFFDL/FES, J. Hodges
- 1 AFFDL/TST, Library

Air Force Test and Evaluation Center, Kirtland Air Force Base, NM 87115

- 1 ATTN: AFTEC-JT

Armament Development and Test Center, Eglin Air Force Base, FL 32542

- 1 ATTN: ADTC/TS

NASA - Ames Research Center, Mail Stop 223-6, Moffett Field, CA 94035

- 1 ATTN: SC, J. Parker

NASA - Ames Research Center, Army Air Mobility Research and Development Laboratory, Mail Stop 207-5, Moffett Field, CA 94035

- 1 ATTN: SAVDL-AS-X, F. H. Immen

NASA - Johnson Spacecraft Center, Houston, TX 77058

- 1 ATTN: JM6
- 1 ES-5

Naval Air Development Center, Warminster, PA 18974

- 1 ATTN: Code 063

Naval Air System Command, Department of the Navy, Washington, DC 20360

- 1 ATTN: AIR-03PAF
- 1 AIR-5203
- 1 AIR-5204J
- 1 AIR-530313

Naval Material Command, Washington, DC 20360

- 1 ATTN: MAT-0331

Naval Post Graduate School Monterey, CA 93948

- 1 ATTN: Code 57BP, R. E. Ball

Naval Surface Weapons Center, Dahlgren Laboratory, Dahlgren, VA 22448

- 1 ATTN: Code G-54, Mr. J. Hall
- 1 Code G-54, Mr. E. Rowe

Naval Weapons Center, China Lake, CA 93555

- 1 ATTN: Code 40701
- 1 Code 408

Commander, Rock Island Arsenal, Rock Island, IL 61299

- 1 ATTN: DRSAR-PPV

Georgia Institute of Technology, School of Mechanical Engineering,
Atlanta, GA 30332

- 1 ATTN: Dr. J. T. Berry

No. of
Copies

To

United States Steel Corporation, Research Laboratory, Monroeville, PA 15146
1 ATTN: Dr. Hsun Hu

Brown University, Division of Engineering, Providence, RI 02912
1 ATTN: Prof. J. Duffy

SRI International, 333 Ravenswood Avenue, Menlo Park, CA 94025
1 ATTN: Dr. D. Shockey

Director, Army Materials and Mechanics Research Center, Watertown, MA 02172
2 ATTN: DRXMR-PL
3 Authors

E
ED